



# A Dual-Polarized, Dual-Frequency, Corrugated Feed Horn for SMAP

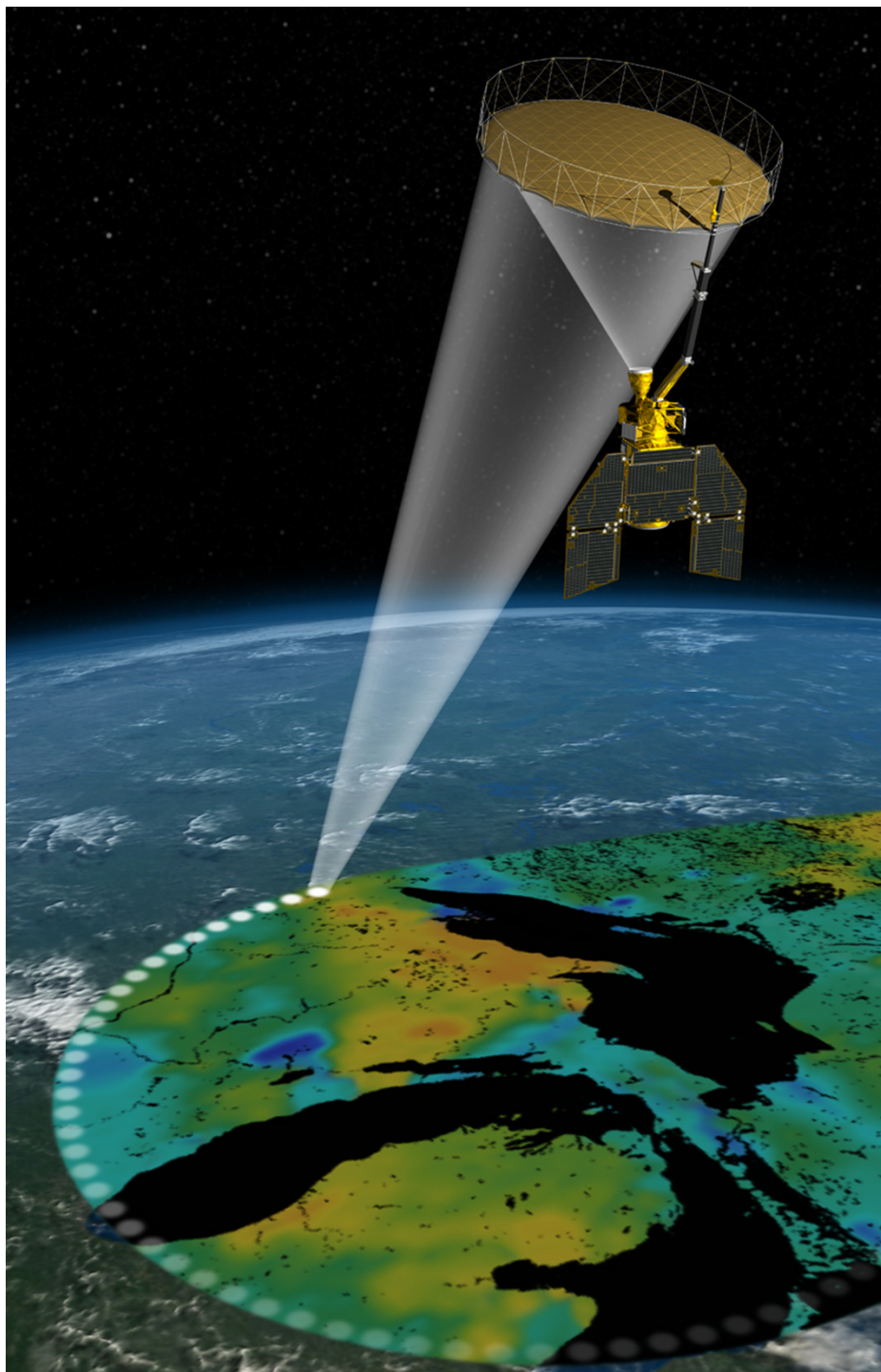
*Paolo Focardi & Paula Brown*

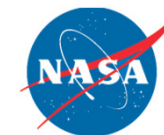
*Jet Propulsion Laboratory, California Institute of Technology*

*2012 IEEE International Symposium on  
Antennas and Propagation*

Chicago, IL  
July 10<sup>th</sup>, 2011

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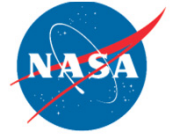




# Outline

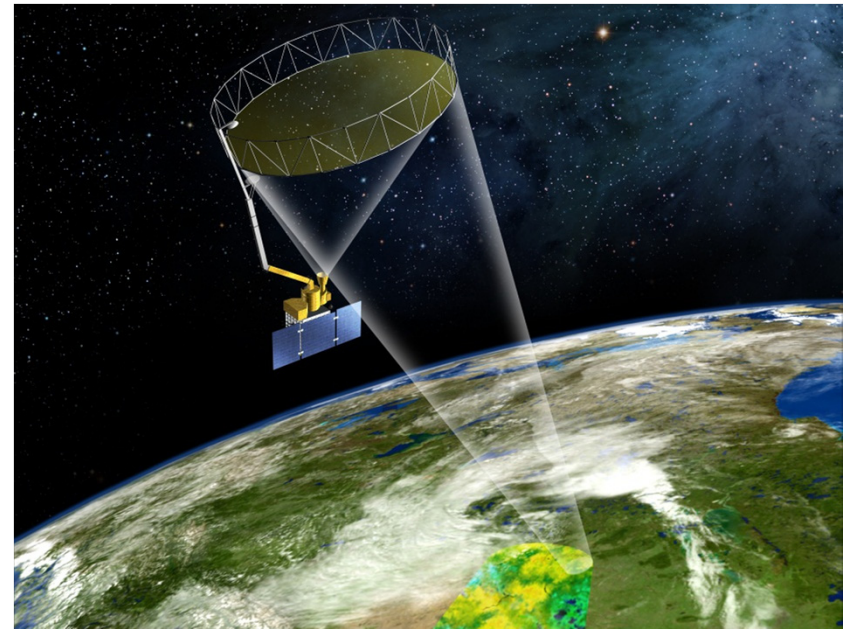
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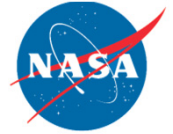
- Overview of the **Soil Moisture Active and Passive (SMAP)** Mission & Instrument
- Basic Layout of SMAP “E” Configuration
- Feed Horn Components
- Major Design Drivers & Requirements
- Thermal and Alignment Tolerances
- OMT Split Design
- SMAP Scale Model
- RL & Pattern Measurements
- Conclusions



# Mission Overview

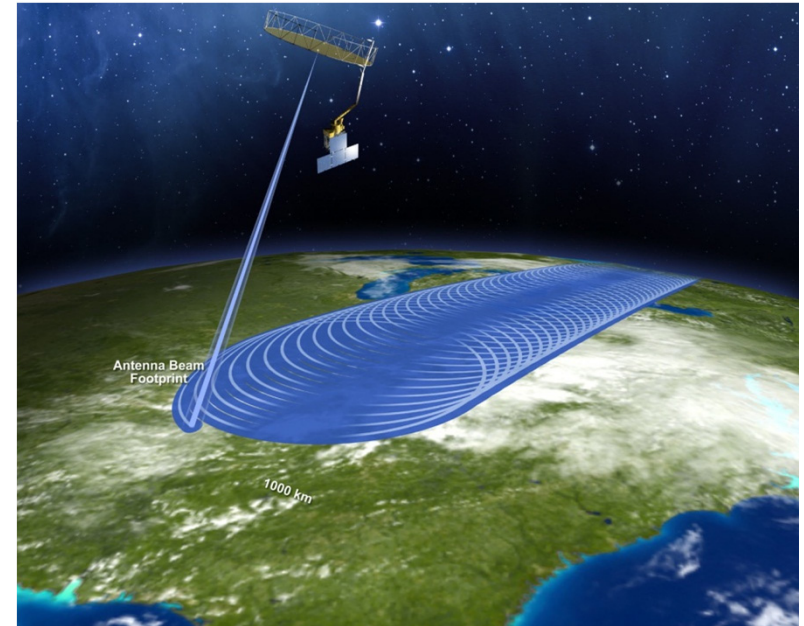
- NASA's **Soil Moisture Active and Passive (SMAP)** mission will measure Earth's soil moisture and its freeze/thaw state over a 3 year period
- Applications:
  - More accurate and longer-term **weather and climate predictions**
  - Earlier **drought warnings**
  - Improved **flood and landslide predictions**
  - Improved **agricultural production predictions**
  - Better understanding of the **global carbon cycle**
- Near-polar, sun-synchronous orbit of **680 km**
- Planned launch date of **November 2014**





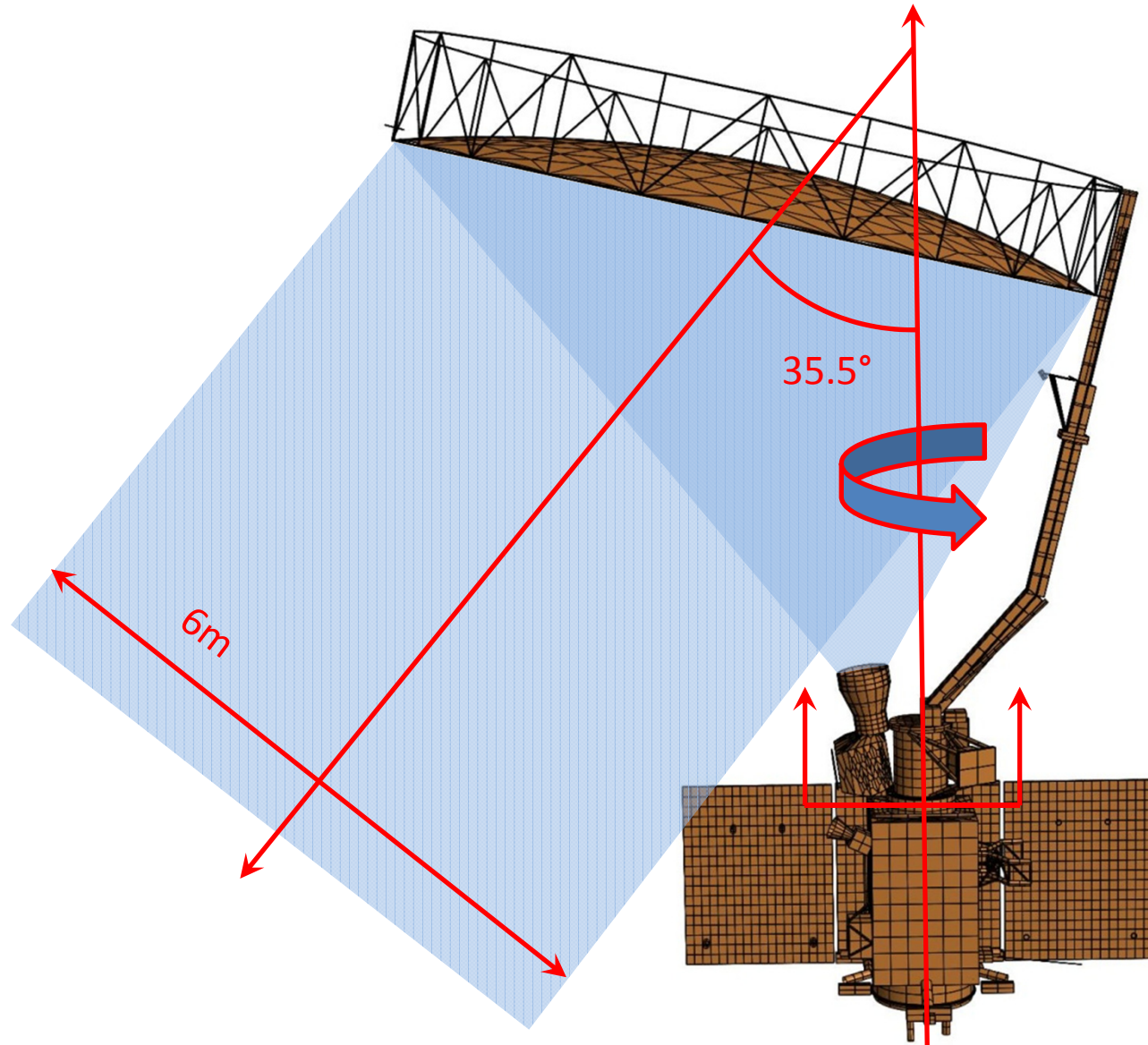
# Instrument Overview

- An L-band Synthetic Aperture Radar (**SAR**) and L-band radiometer (**RAD**) share an **offset 6-m deployable mesh reflector** and feed
- The antenna **boresight** beam is pointed **35.5°** off nadir
- The instrument **spins** at approximately **14.6 RPM** around the nadir axis
- The result is a **1000-km swath** on the ground
- The radiometer data is more accurate than the SAR data, but has a spatial resolution of about 40-km; the SAR spatial resolution is 1 – 3 km

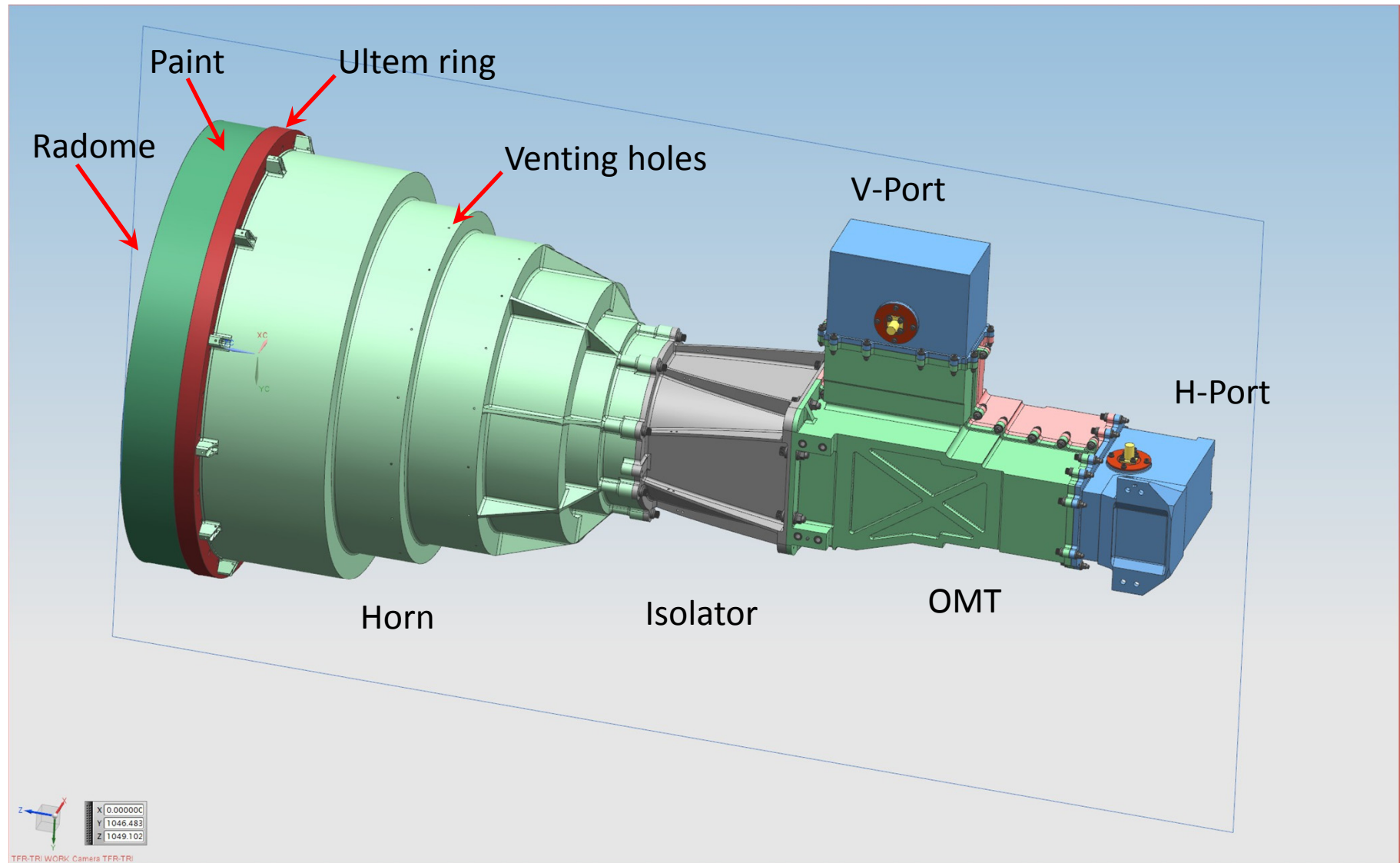




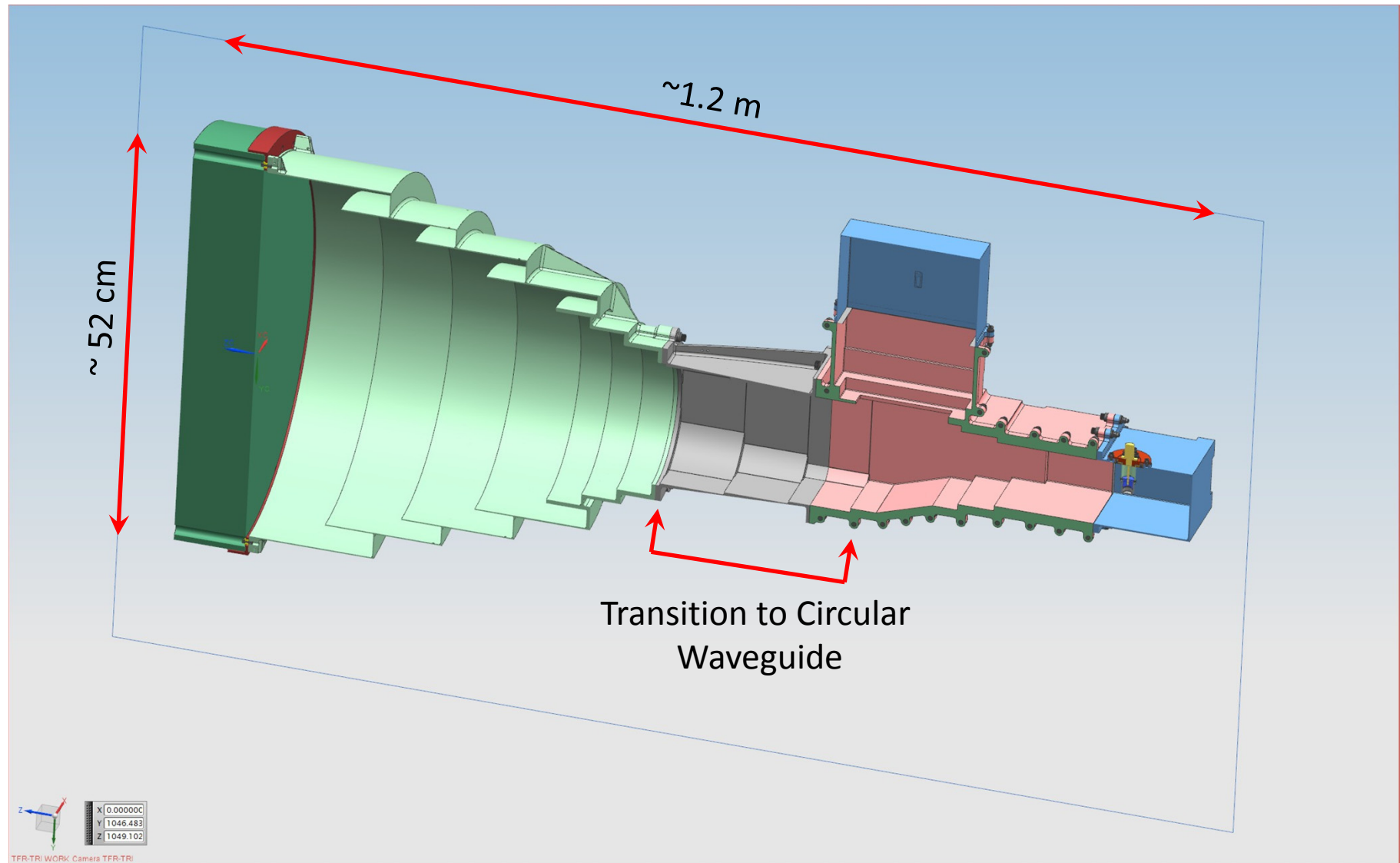
# SMAP "E" Configuration

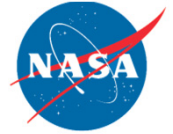


# Isometric View

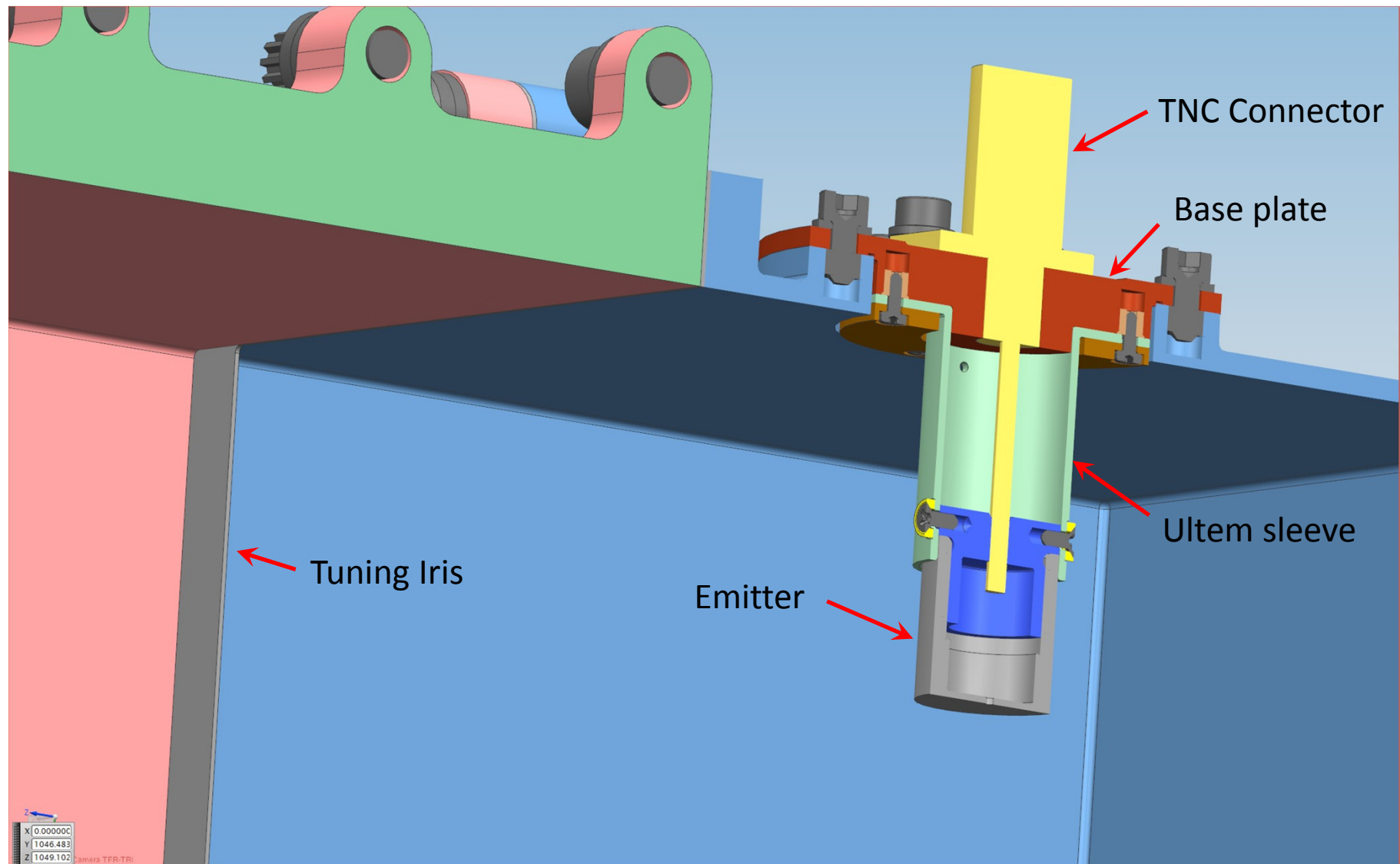


# Cut-away Isometric View



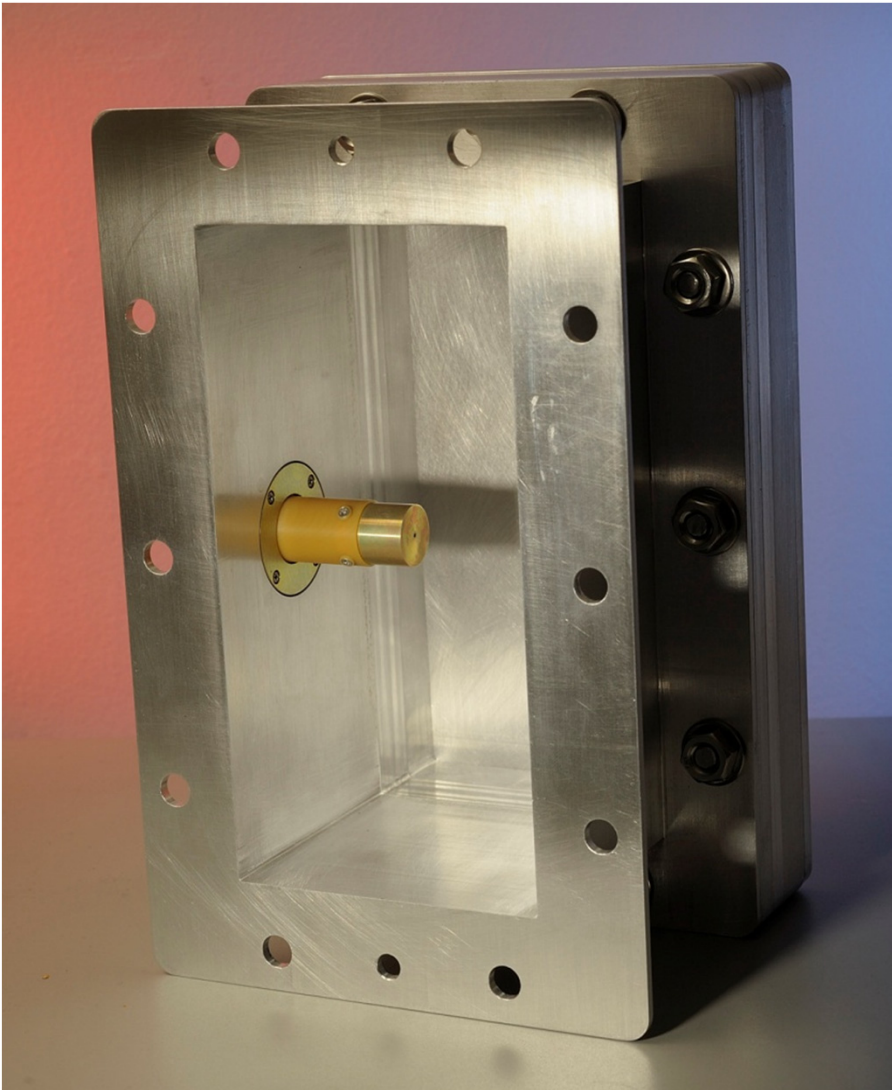


# Waveguide to Coax Adapter (WCA) Detail

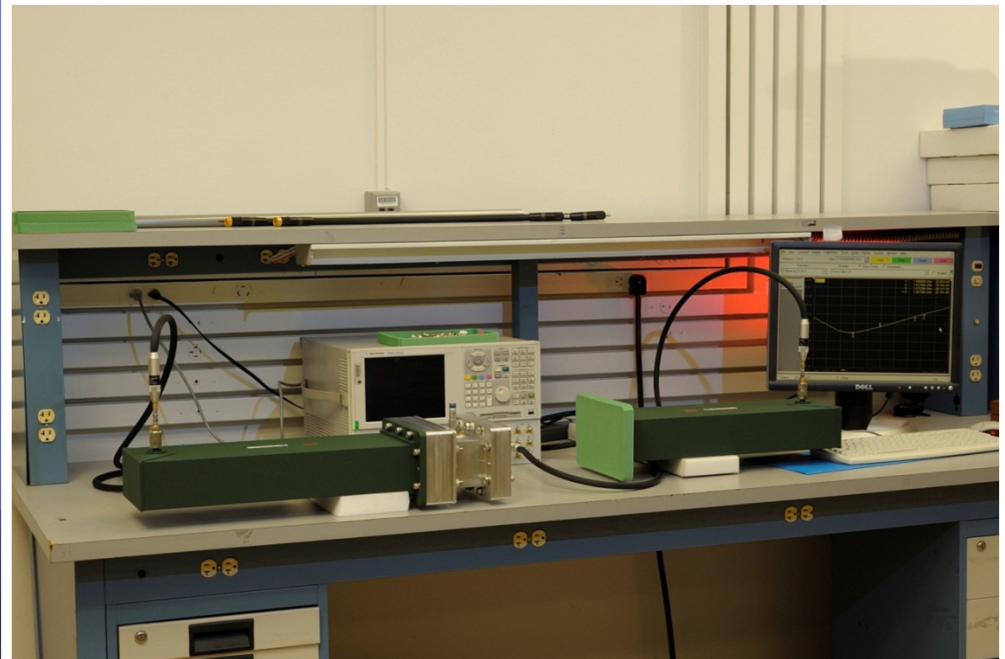


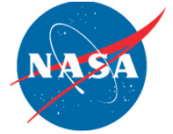


# WCA Prototype Measurement



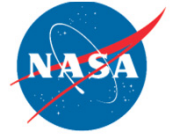
**WCA Prototype being tested with  
WR650 Cal Kit**



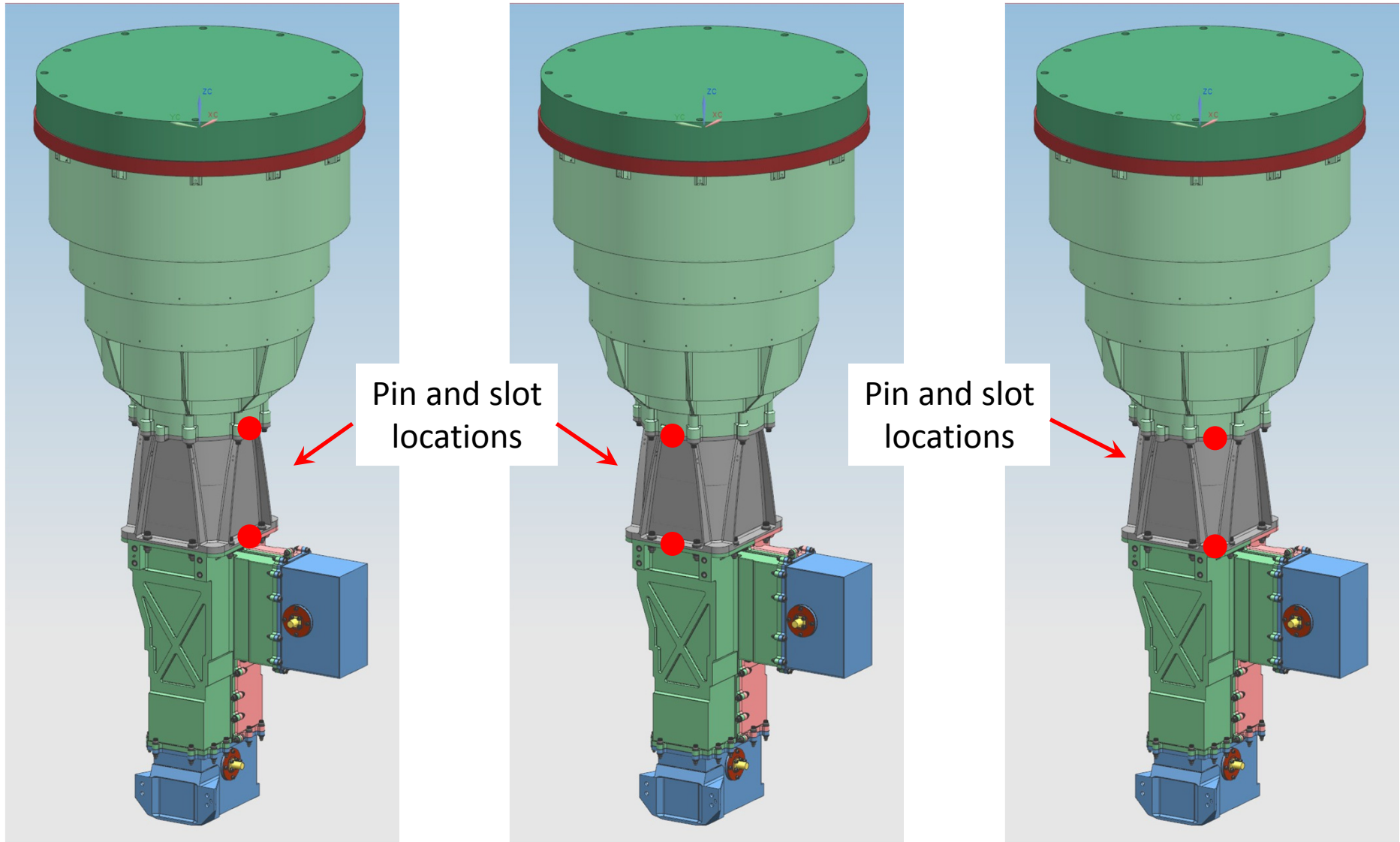


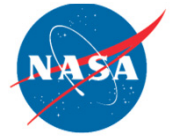
# Major Design Drivers & Requirements

- Combined SAR & RAD RF bandwidth of 16%
- Radiometer beamwidth & main beam efficiency
  - RAD Beamwidth between  $2.29^\circ$  and  $2.5^\circ$
  - RAD MBE  $> 87\%$
  - SAR Beamwidth  $< 2.8^\circ$
- Radiometer antenna pattern stability
  - RAD Earth Lobe power  $< 3\%$
  - RAD off-Earth Lobe power  $< 10\%$
- SAR gain and gain stability
  - SAR Gain  $> 35.55$  dBi
  - SAR Gain stability  $< 0.07$  dB
- SAR pointing stability
  - $50^\circ \pm 40^\circ$  in Elevation
  - $0^\circ \pm 10^\circ$  in Azimuth

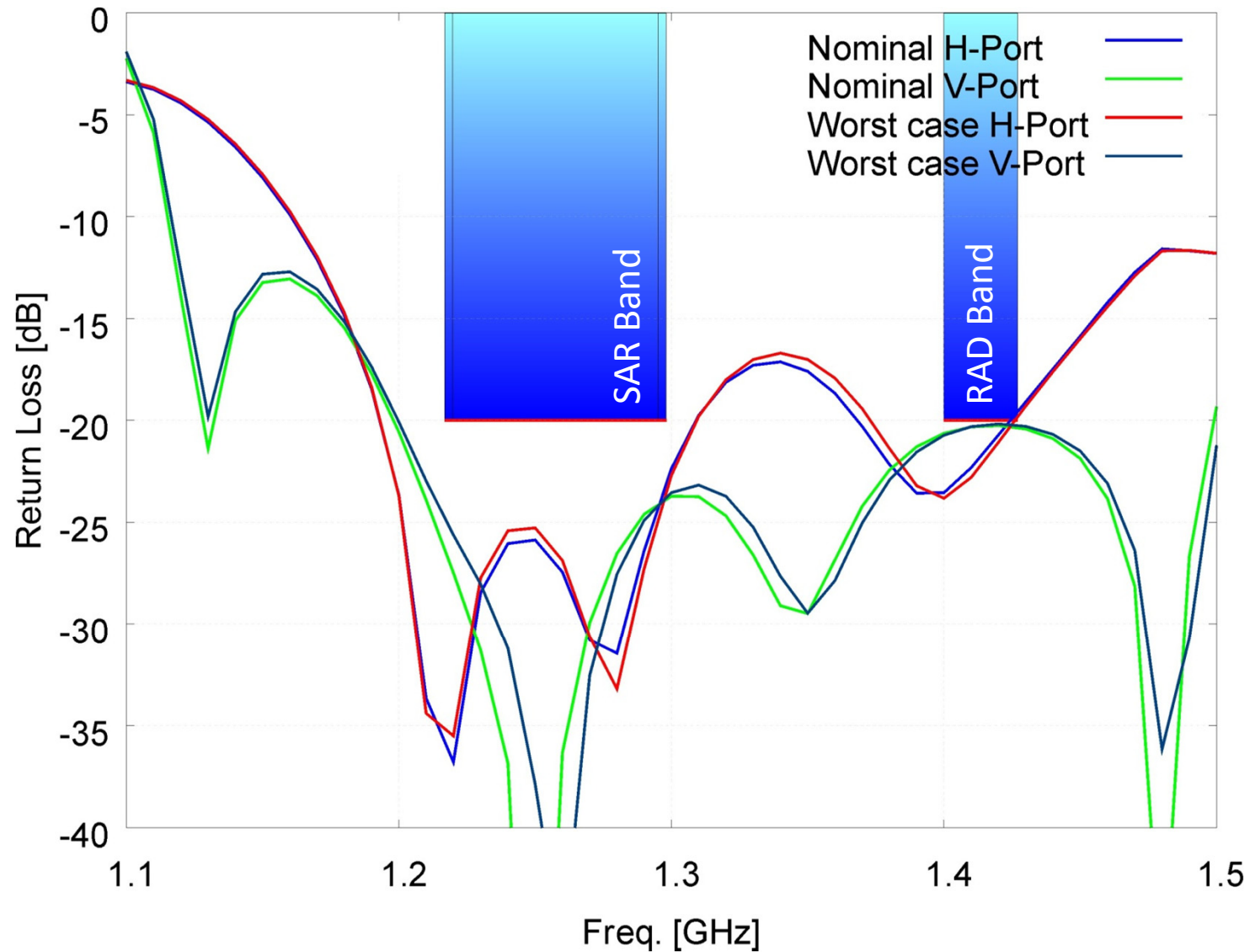


# Thermal & Alignment Tolerances

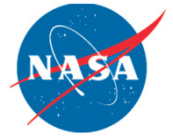




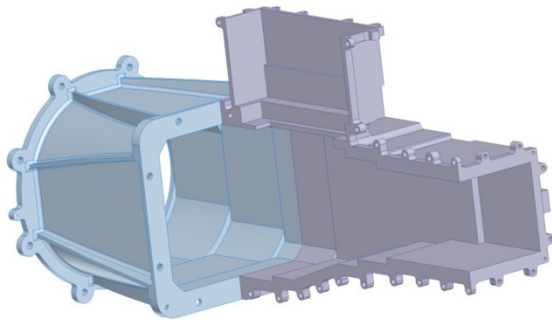
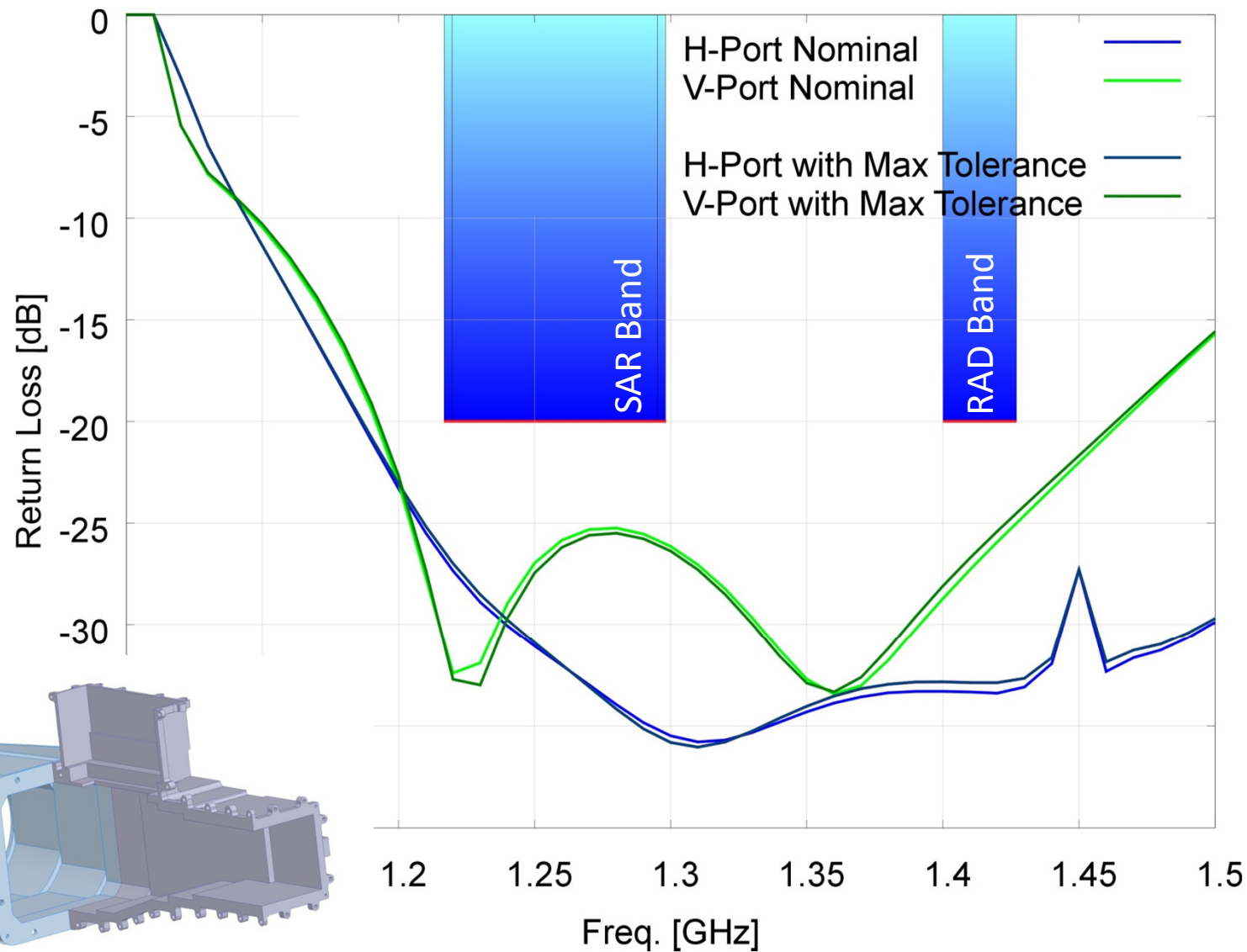
# Worst Case Thermal + Alignment Tolerances



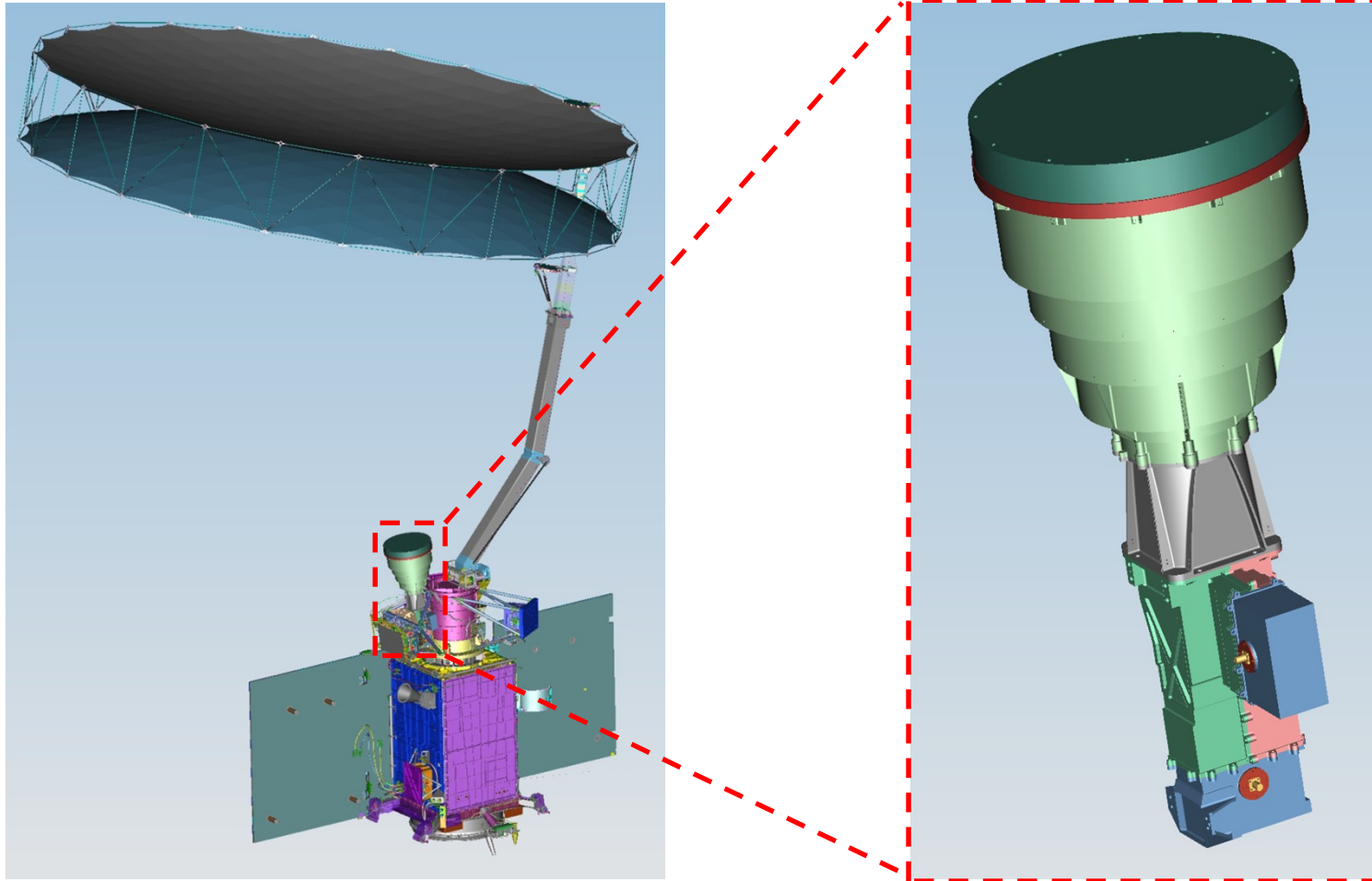




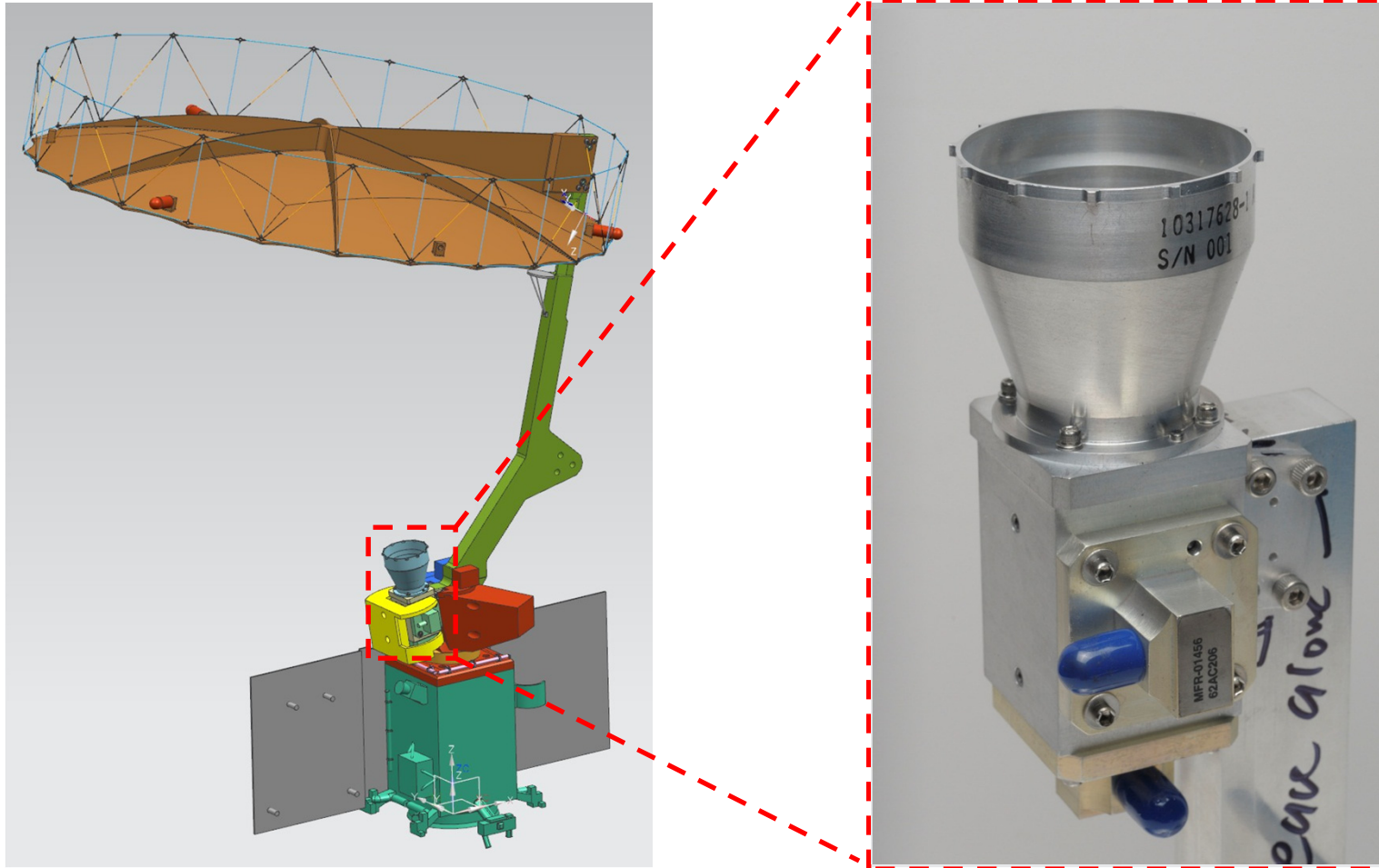
# Split OMT Tolerance Performance

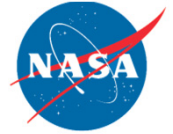


# SMAP Flight Model



# SMAP Scale Model

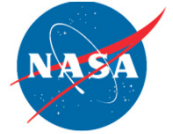




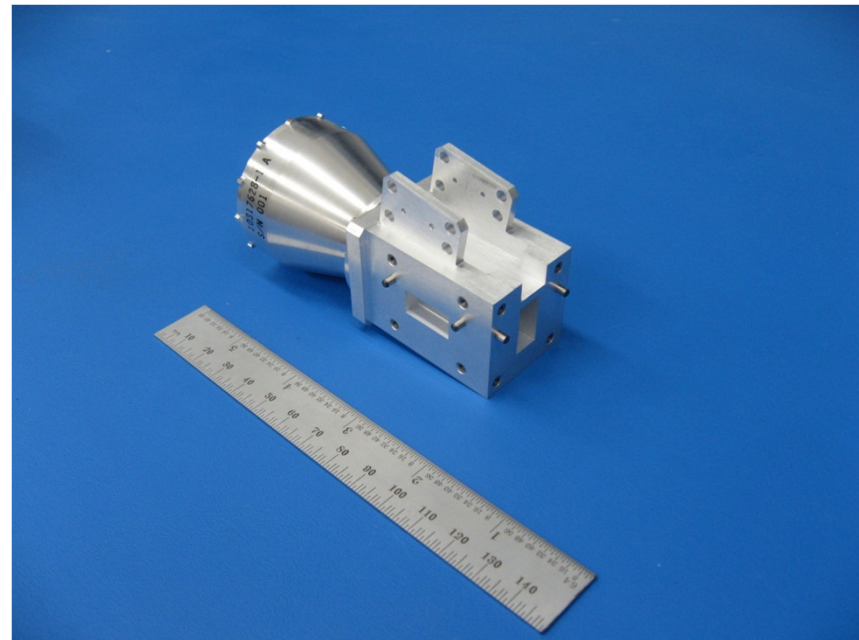
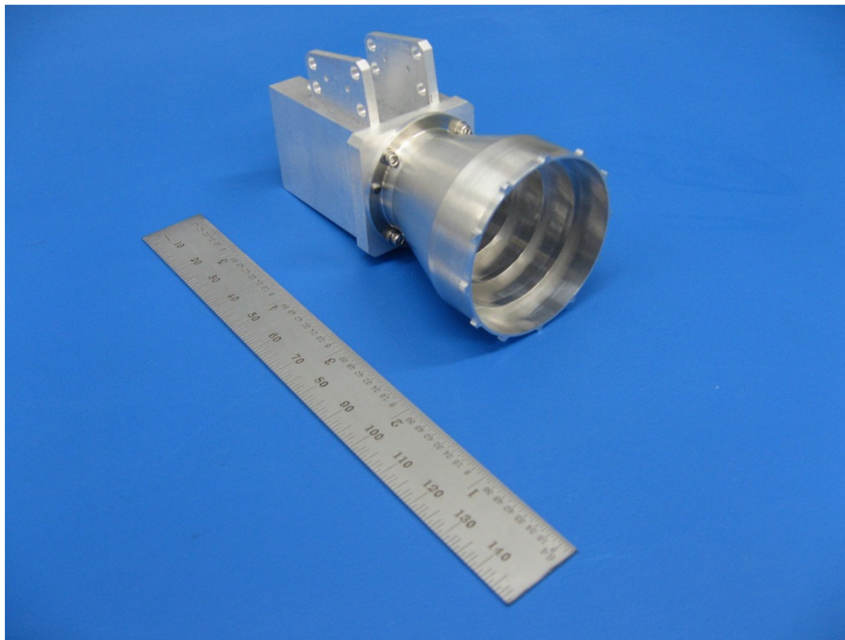
# SMAP Complete Scale Model

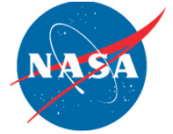




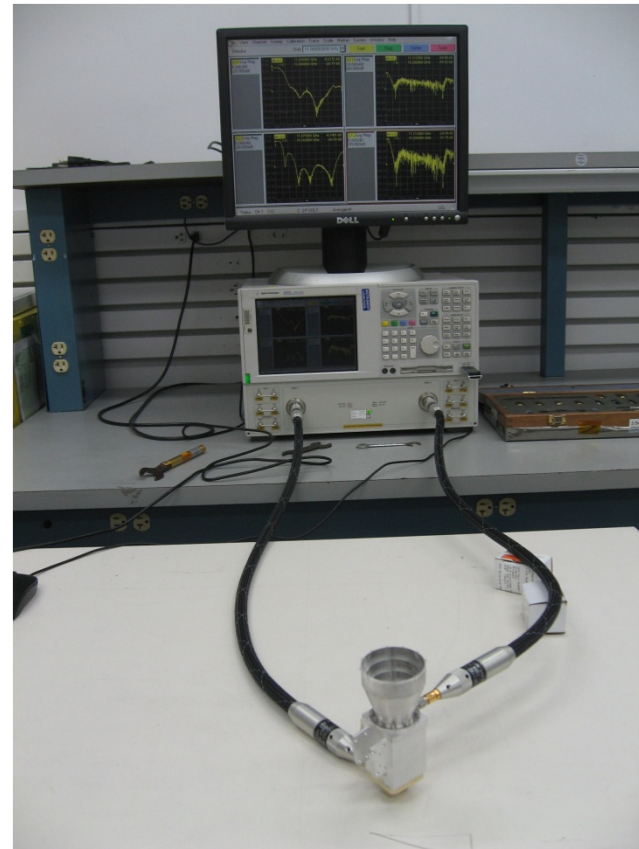
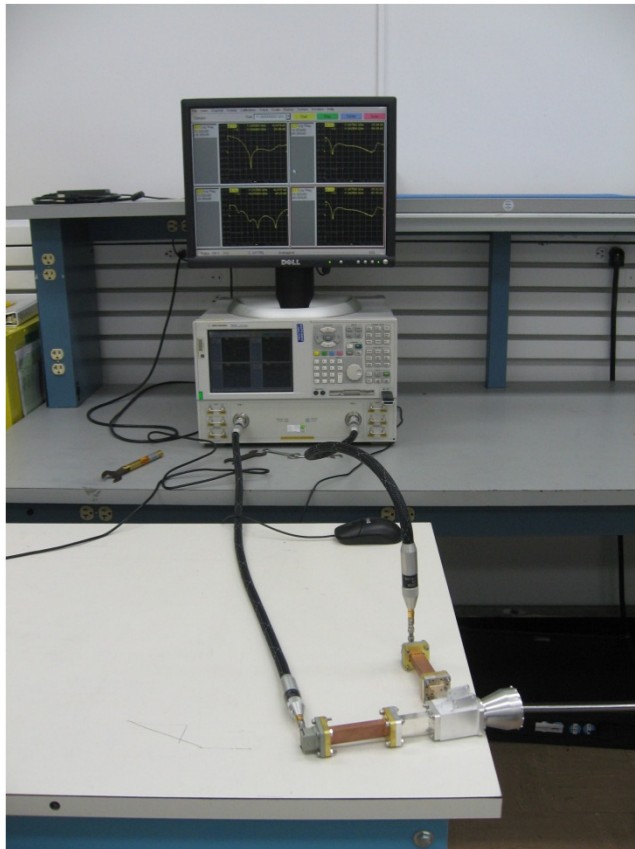


# Scale Model Feed



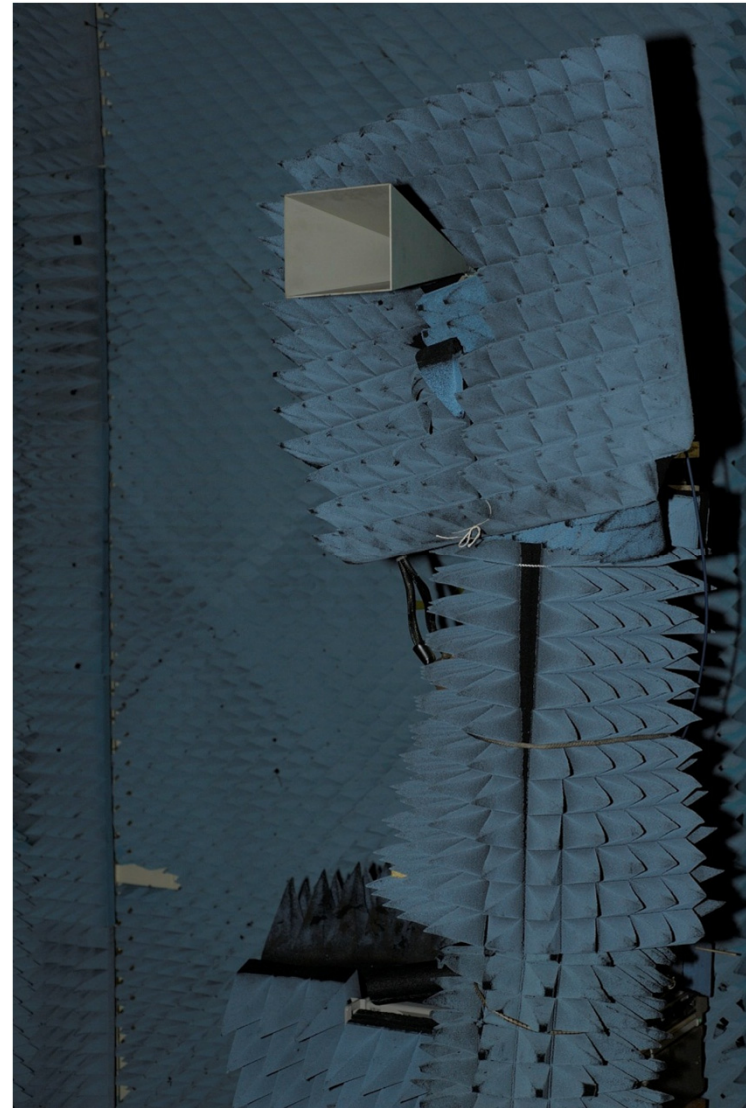
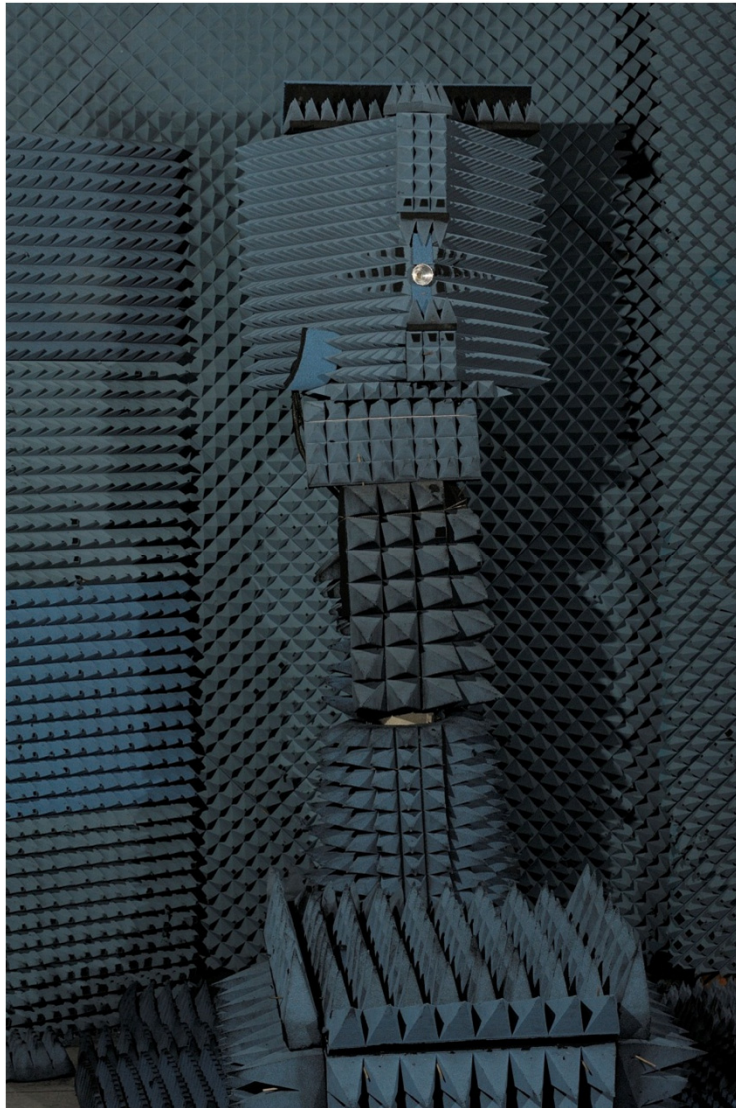
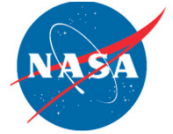


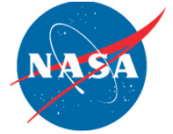
# Return Loss Measurements





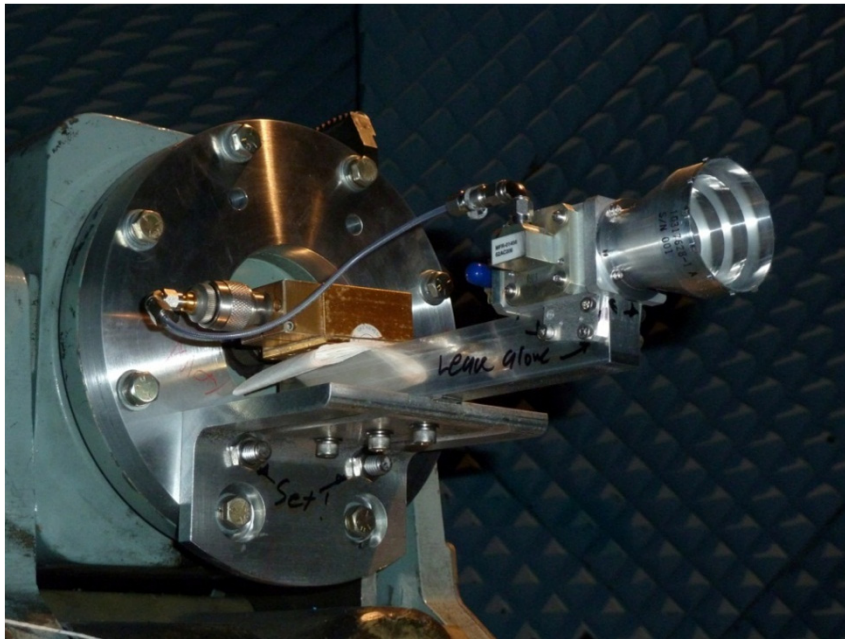
# Radiation Pattern Measurement



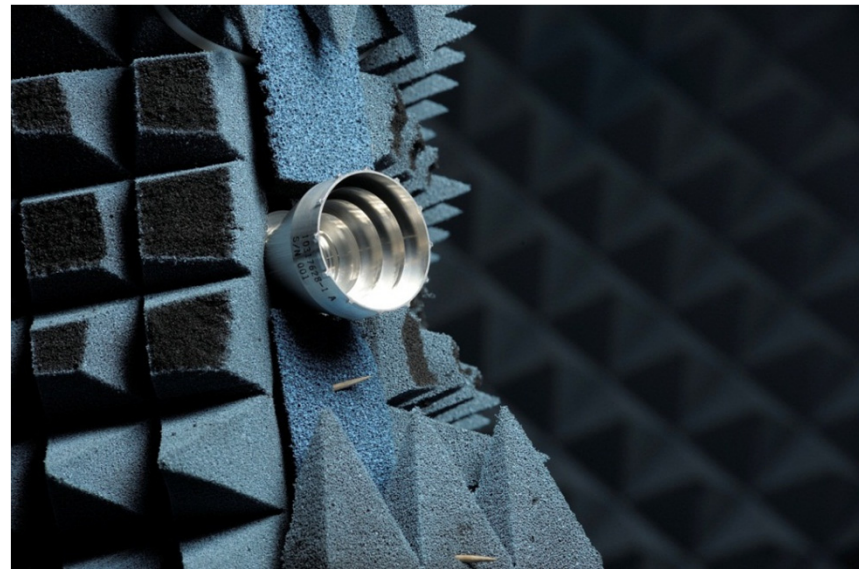


# Close-up of the Scale Model Feed Horn

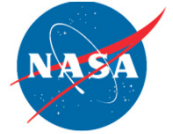
**Scale Model Feed Horn on Antenna Range Positioner**



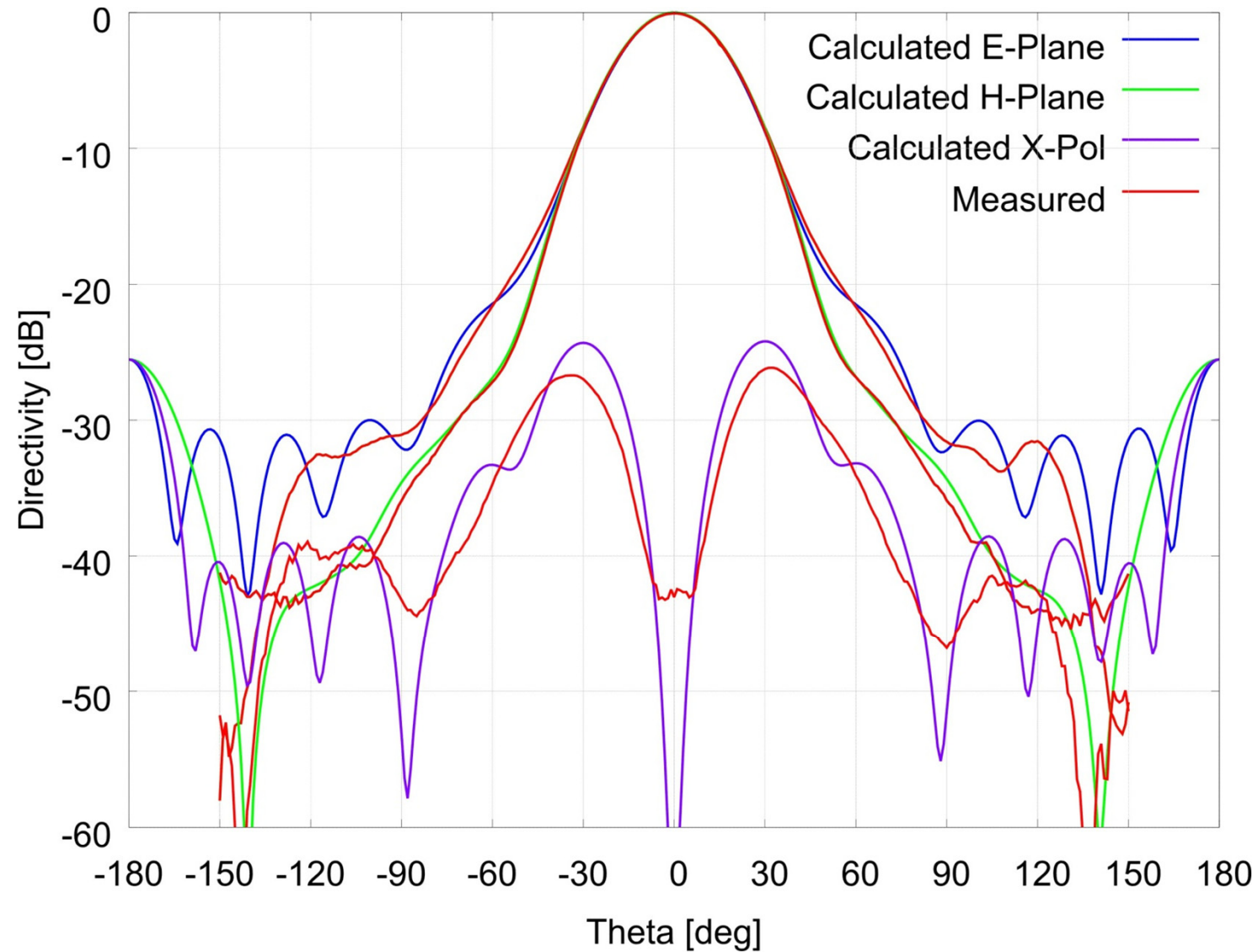
**Scale Model Feed Horn with Absorber in Place**

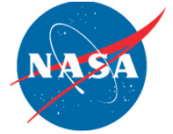




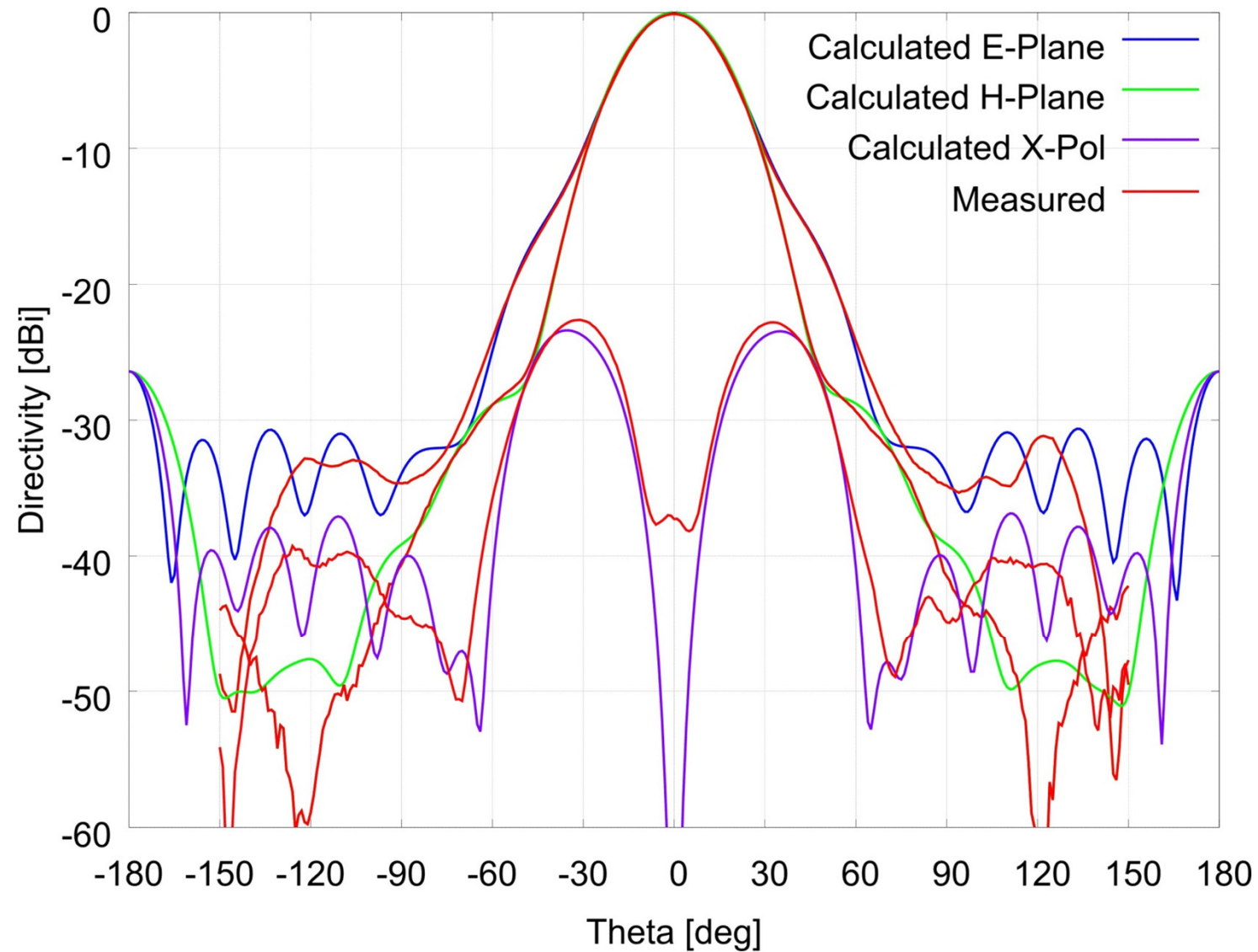


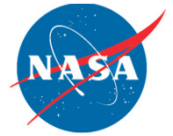
# SAR V-pol Scale Model Pattern



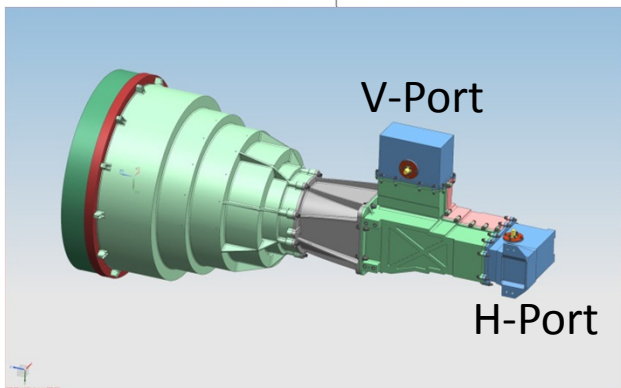
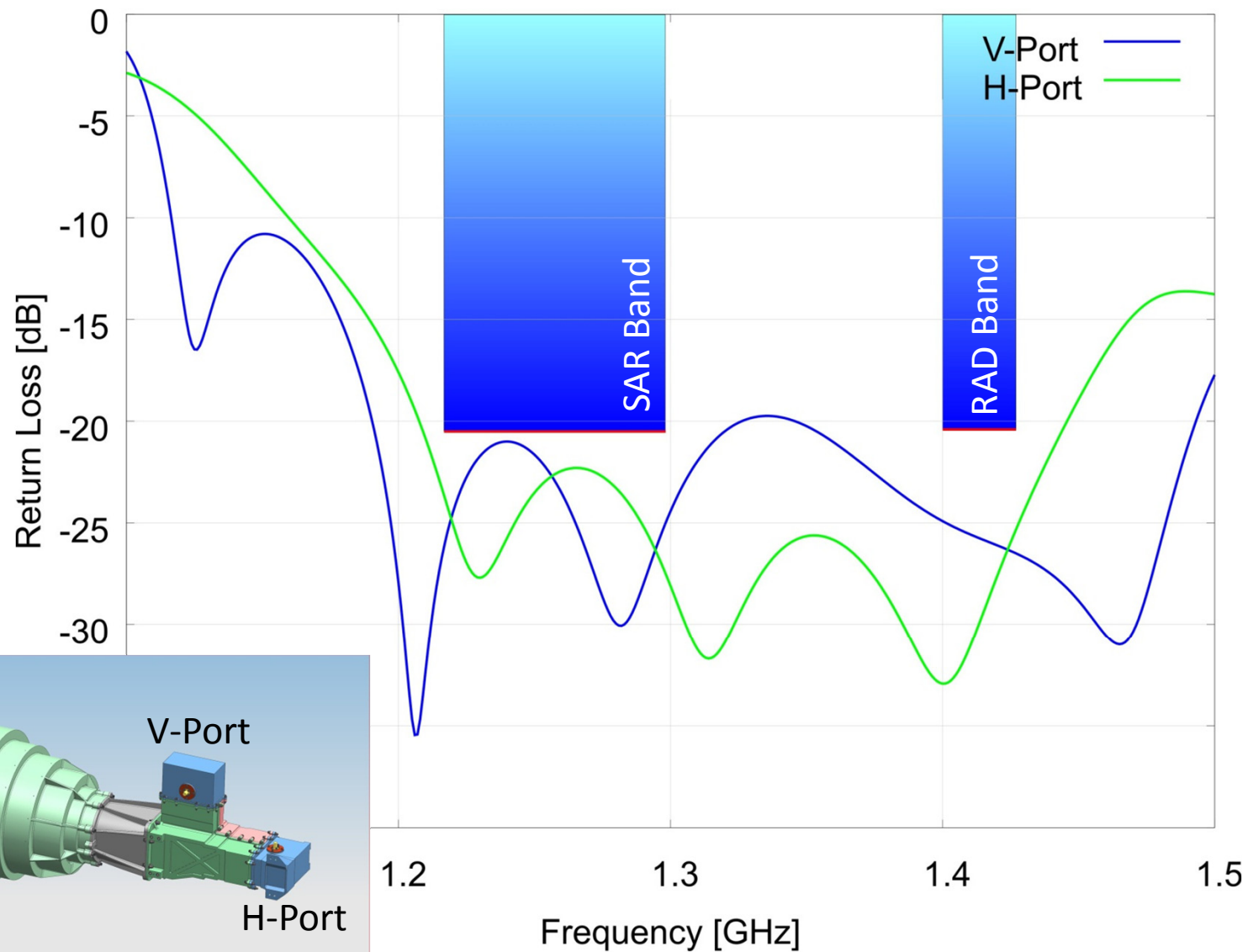


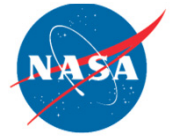
# RAD V-pol Scale Model Pattern



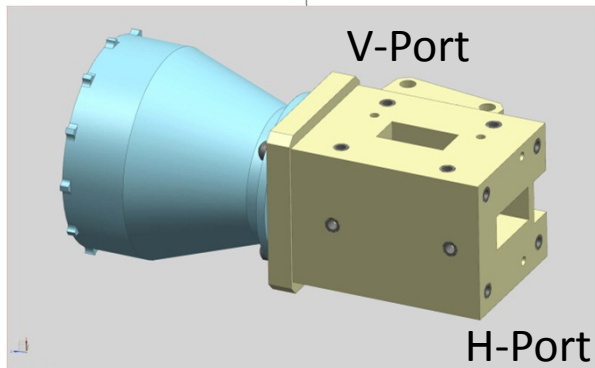
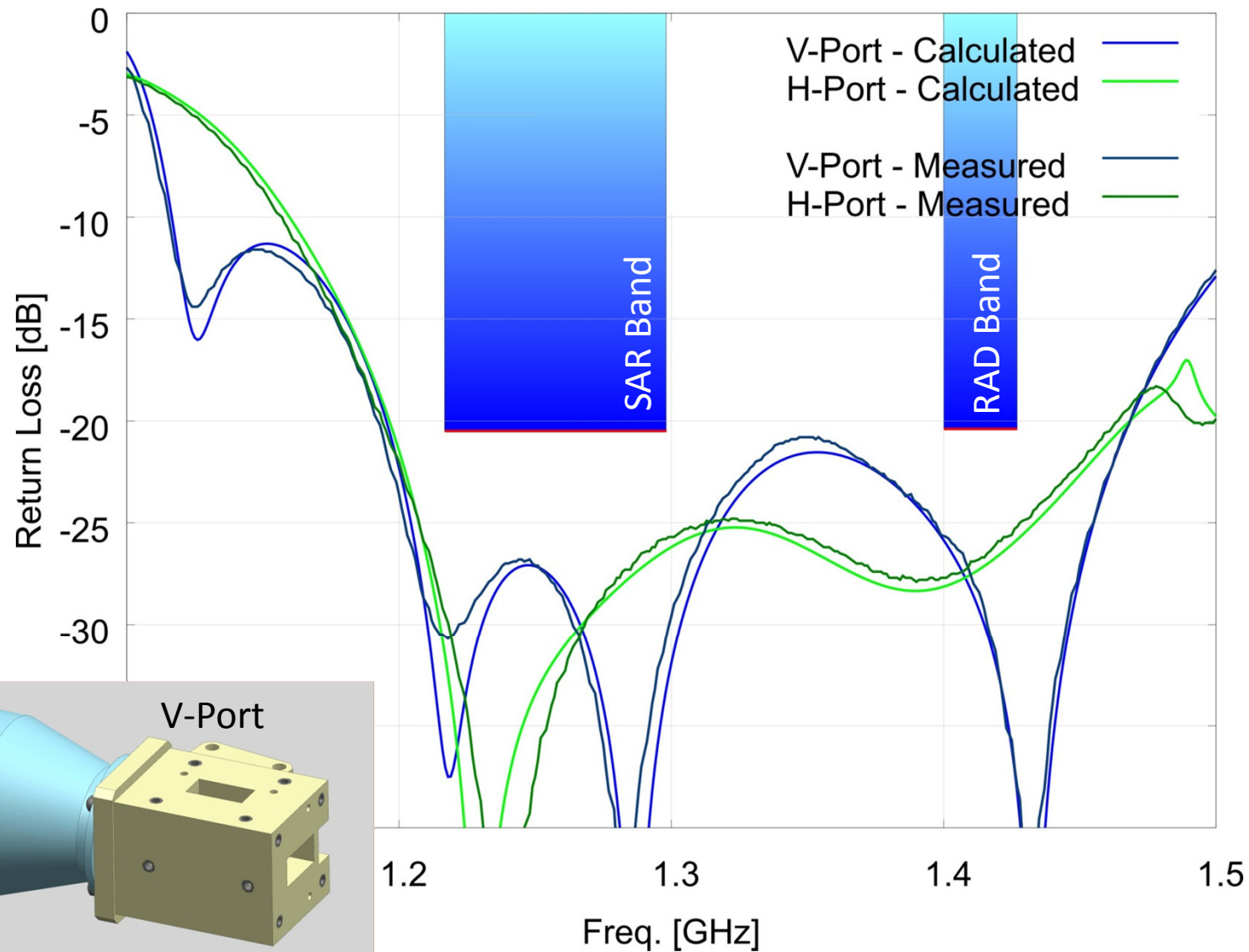


# Flight Feed Horn RL

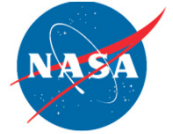




# Feed Horn RL into SM OMT



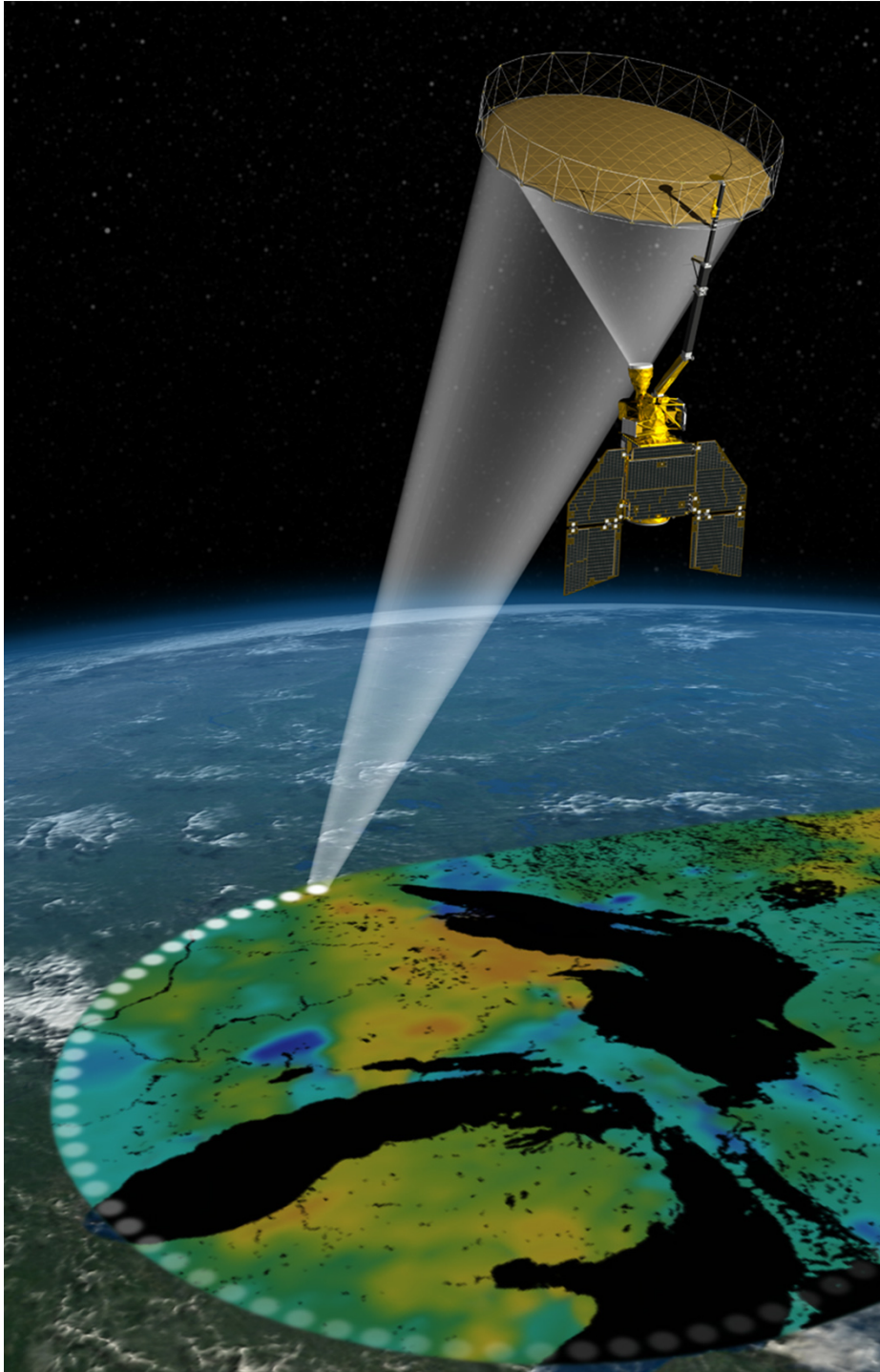




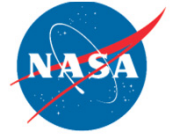
# Conclusions

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- A dual-polarized, dual-frequency, corrugated feed horn for SMAP was designed and meets all mission requirements
- A scaled model of the feed was fabricated and tested showing an excellent agreement with predicted performance
- These results along with the “tunability” built-in into the flight model make us feel confident that the flight model will meet all mission requirements



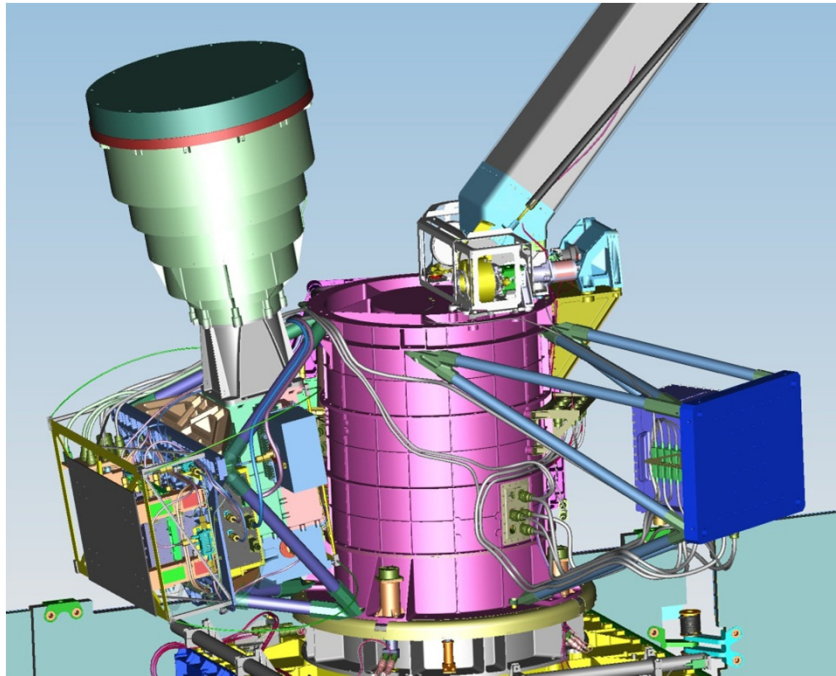
**Thank you!**



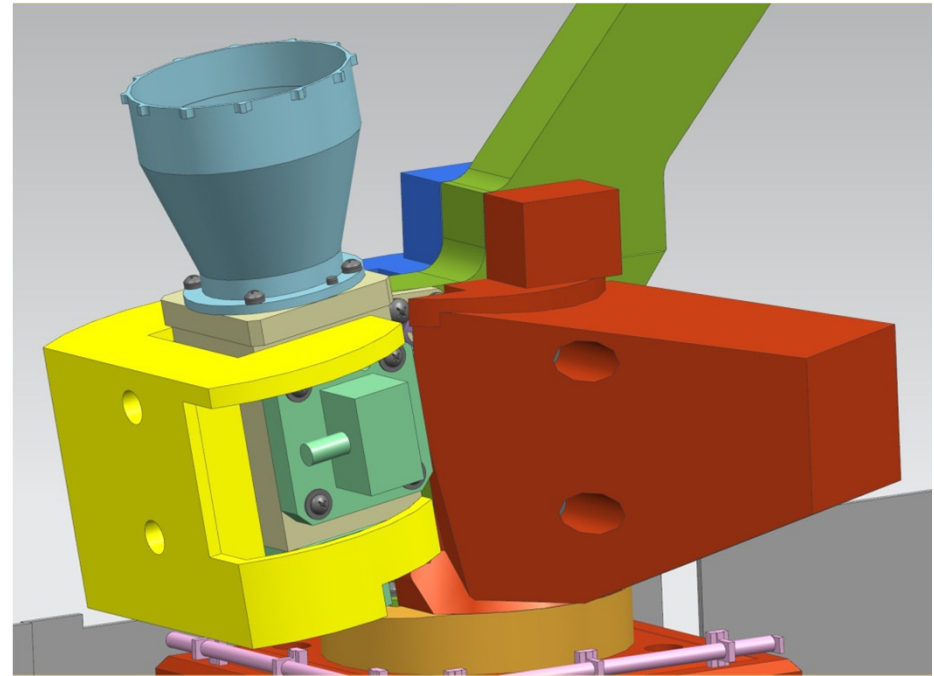
# Back-Up

# SMAP Instrument

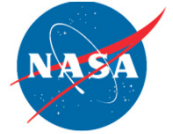
## Flight Model



## Scale Model

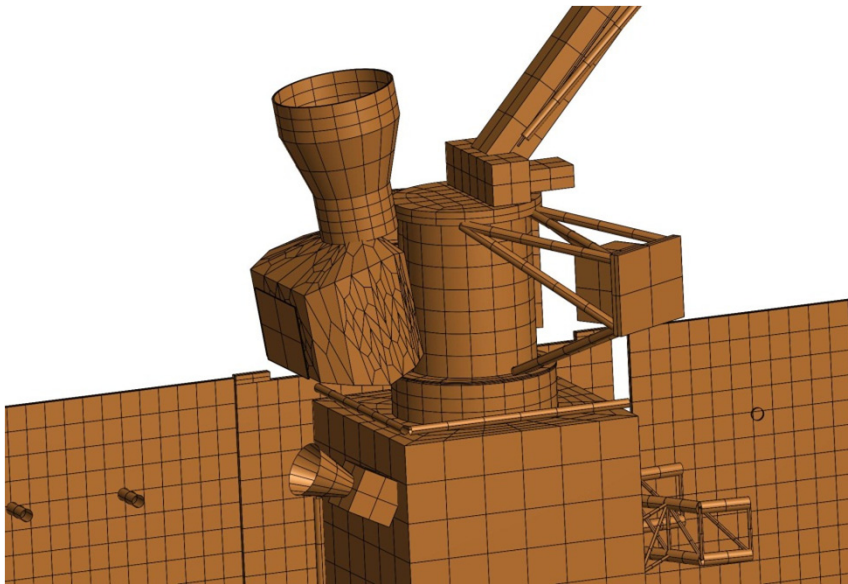




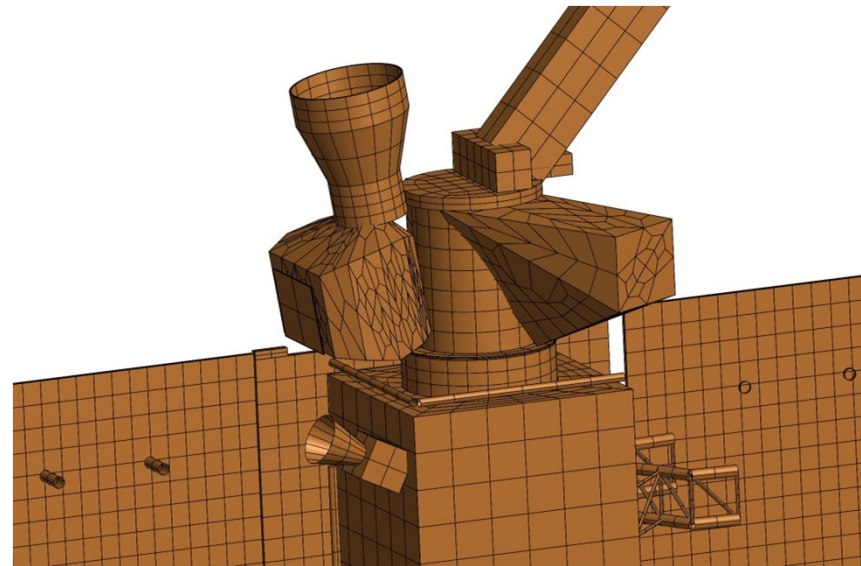


# RF Models

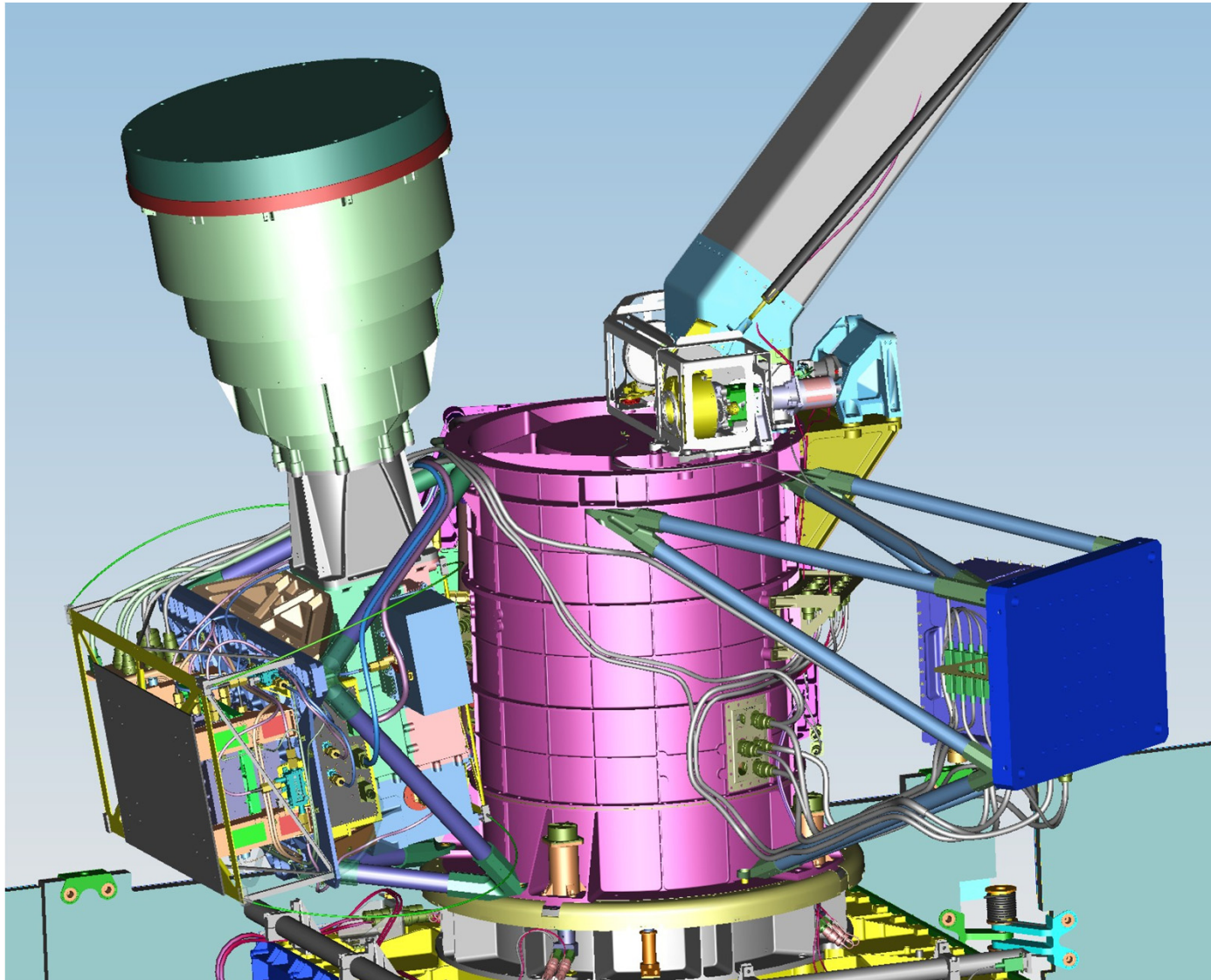
## Flight RF Model

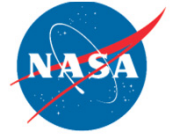


## Scale Model RF Model

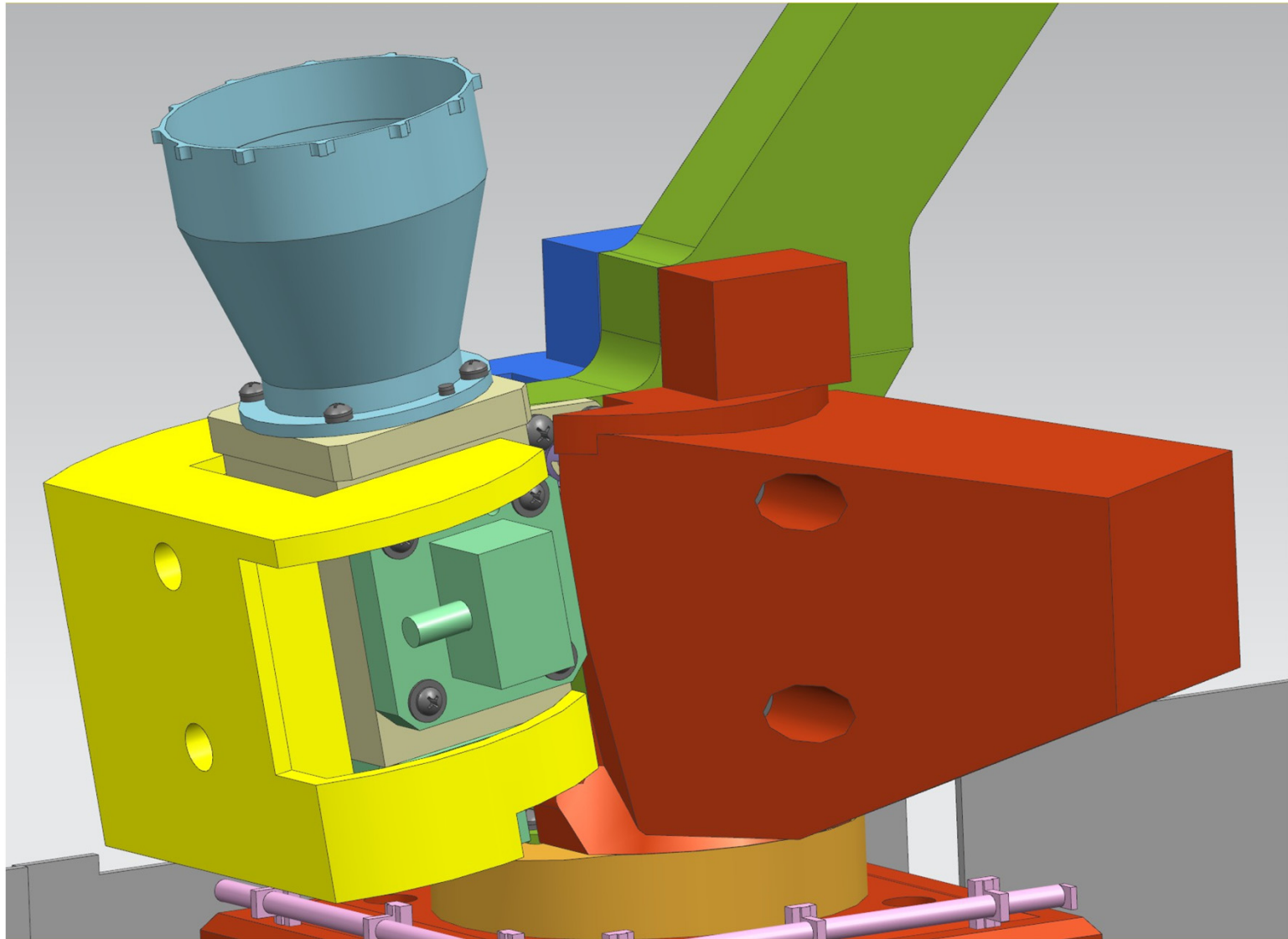


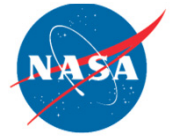
# SMAP Flight Instrument





# SMAP Scale Model Instrument





# Radome

